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Title: PLASMA FLOW SWITCH AND FOIL IMPLOSION EXPERIMENTS
ON PEGASUS II

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PLASMA FLOW SWITCH AND FOIL IMPLOSION EXPERIMENTS ON PEGASUS II

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Introduction

Pegasus II is the upgraded version of Pegasus, a pulsed power machine used in the Los Alamos AGI-X (Above Ground I Experiments) program. A goal of the program is to produce an intense (>100 TW) source of soft x-rays from the thermalization of the kinetic energy of a 1 to 10 MJ plasma implosion. The radiation pulse should have a maximum duration of several 10's of nanoseconds and will be used in the study of fusion conditions and material properties. The radiating plasma source will be generated by the thermalization of the kinetic energy of an imploding cylindrical, thin, metallic foil. This paper addresses experiments done on a capacitor bank to develop a switch (plasma flow switch) to switch the bank current into the load at peak current. This allows efficient coupling of bank energy into foil kinetic energy.

Figure 1 is a drawing of the Pegasus II facility. Pegasus II machine parameters include a stored energy of 4.3 MJ at 100 kV, a system inductance of 10 nH and current capability of 18 MA. This quadruples the energy of Pegasus I at this voltage. The upgrade was accomplished by replacing the capacitors rated at 10 kJ stored energy at 60 kV with capacitors rated at 30 kJ at 80 kV. The new capacitors have a current capability of 250 kA/capacitor and can stand up to 20% reversal at full charge for a rated lifetime of over 3000 shots. To stay within this voltage reversal specification, series fuses are employed to shut off the current after peak current. The bank itself is composed of two halves charged to opposite polarity. Each half has four modules with eighteen capacitors each. The modules are placed around a radial transmission line with the load in the center of the line. Deflection switches¹ which form an annular deionization jet that penetrates the polyethylene switch insulation are used to switch the bank. The facility has been used in "direct drive" z-pinch implosions of thin aluminum foils, high magnetic field diffusion experiments, pulse sharpening and switching experiments using a plasma flow switch, and most recently, in liner experiments where the load is an aluminum cylinder with a 0.4 mm thick wall.

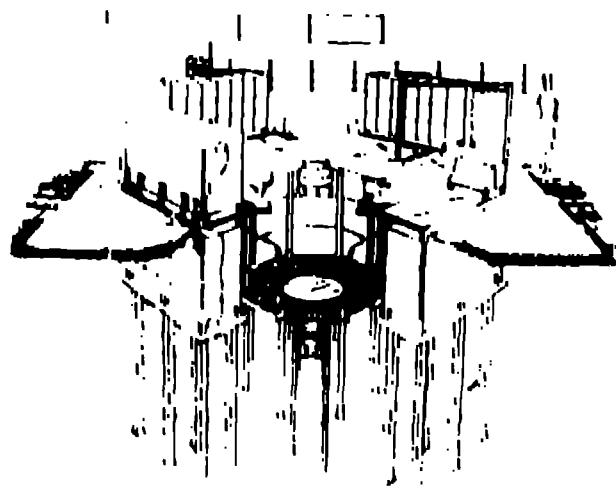


Fig. 1. Pegasus II facility.

Experiment Results

The purpose of the plasma flow switch (pls) experiments presented here is to develop a switch that demonstrates switching into a high-inductance dummy load. On previous experiments done on Pegasus I, we have observed efficient switching of all of the drive current into a static load at a radius of 5 cm (the foil radius). However, switches driving implosion loads have not been able to develop voltages necessary to drive good implosions (V_{imp} > 100 kV). Speculation is that the switch cannot support a high switching voltage because of plasma entry in the power flow channel or bridging the load slot. The experiments done on Pegasus II use a thick copper cylinder of 1 cm radius as the load. Figure 2 is a drawing of the Pegasus II power flow channel showing the location of the pls in relation to the load slot. The pls is made of two components: a thin aluminum "bridge" that shorts the power flow channel and a mylar barrier film located just downstream of the aluminum. When the bank is discharged, the aluminum becomes a plasma that conducts the current of the capacitor bank. The plasma then starts to accelerate down the power flow channel via the $B \times B$ force acting on the current-carrying plasma. The barrier film inhibits the motion of the aluminum plasma until it burns away after about 500 ns (depending on film thickness). The assembled plasma then moves down the channel with a fairly well defined front. As the plasma slug crosses the load slot (near peak bank current), the current path then includes the load. A simple model for the time to switch the current into the load slot is velocity of the pls divided by the width of the load slot. In our experiments, this number is 200 picoseconds.

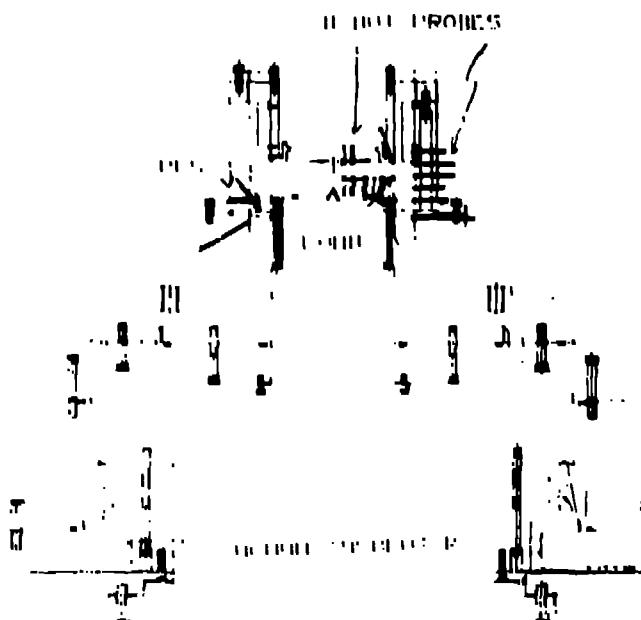


Fig. 2. Pegasus II load chamber.

We are developing the plasma flow switch as opposed to the more commonly used plasma erosion switch because of the relatively

as rise times longer than current drive current in the AdX segment at low Aridity. In the capacitor bank and explosive flux compression experiments, in both of these cases, the conductance was too long for a demonstration experiment switch to support the drive current before opening. Plasma driven switch have sustained the driving current for up to 10 μ s after opening and our requirements relate to a switch that will conduct many mega amperes for at least 10 μ s.

As mentioned by Dr. Traylor, the pbs actually steepens the上升 front of the current pulse moving down the power flow channel. This is shown in Fig. 3 where current vs. time at different axial positions is shown. Note that the waveform does steepen in time and that there is a "foot" on the rising portion of the current wave. This foot is greatly reduced in amplitude as observed by PFC probes located in the load slot. The amplitude of this foot and its relation to pbs initiation and pbs mass distribution is under investigation. The mass distribution of the pbs used in Pegasus up to this point is a $1/r^2$ distribution to match the magnetic pressure profile. This has been done on Pegasus I by using a chordal wire array combined with a constant thickness mylar barrier film. The newer pbs shots fired on Pegasus II have used a mass graded aluminum foil instead of the wire array. Results will be compared between these two methods of forming the switch plasma in future experiments.

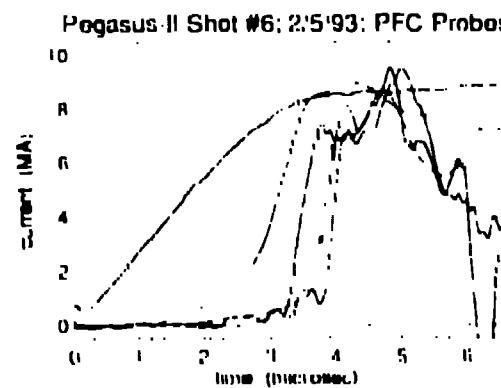


Fig. 3. Current pulse shape as plasma moves down pbs.

The successful results of the "Clock Line" series of experiments done on Shiva Star are well known. The main difference between those experiments and the ones performed on Pegasus have been current levels. Pegasus I operated at about 1.1 MV with a negative 1 dot at switch time whereas Shiva Star operated at about 1.2 MV with positive 1 dot at switch time. Pegasus II is much closer to the parameters of the Shiva Star experiment, with current levels approaching 1.9 MVA at switch time and switch time while 1 dot is still positive.

Figure 4 shows current switch curves for two pbs experiments on Pegasus II. These two experiments had pbs masses of 100 me and 50 me. The switch occurred at lower current with the lower mass because of the reduced time to travel to the load slot. Nevertheless, the smaller mass produced better switching as shown in the cold load in Fig. 4 or 4.3 (b), where the current switched on the cold load at -3 μ s is shown. Note that the current for the 100 me and the 50 me is at a higher off switch current level. These steps are associated with the loaded load on the load slot as a fast rate exceeds some threshold value. Another consequence of the lighter pbs is that 1 dot is more positive than with the more massive switch since switch occurs earlier in

the fall of the current. Figure 5 shows the inferred voltage sustained across the load slot by the pbs during the switching event. Note the significant improvement of the Pegasus II experiments over the Pegasus I results. It should be noted that all of these experiments were performed with the "plasma trap" developed for Pegasus I experiments. This trap was included to prevent a layer of plasma from degrading the load slot and results obtained on Pegasus I supported this design. Calculations have shown that as current levels increase, the effect of the trap may actually be harmful. Future experiments are planned to investigate this effect.

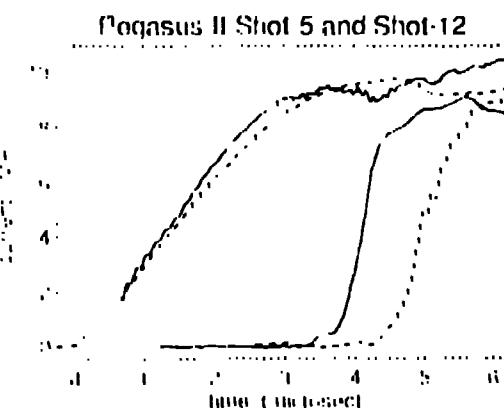


Fig. 4. Bank current and switched current for 50 me and 100 me pbs experiments.

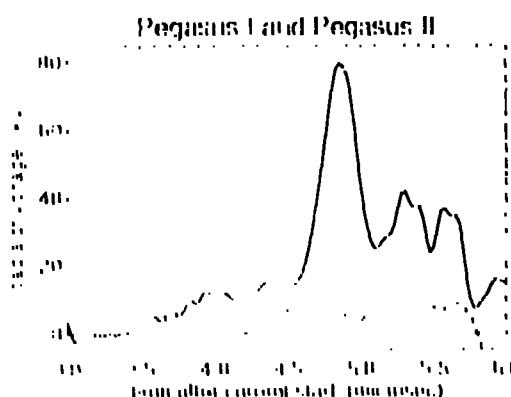


Fig. 5. Switch voltage across load for Pegasus II and Pegasus I pbs experiments.

Summary

The preliminary experiments performed on Pegasus II using plasma flow switch have been encouraging. We are studying the switch on the all-silicon diode load so that switch behavior is the only variable. The parameters of Pegasus II seem better matched to continue the switching performance of the pbs on Pegasus I. Results to date have shown that all of the driving current can be switched into a high inductance stationary load in roughly 10 μ s. Experiments are continuing to optimize the switching characteristics of the plasma flow switch.

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