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MEASUREMENT AND ANALYSIS OF THREE 1.5-GPA Title: SHOCK-WAVE PROFILES IN COPPER Richard H. Warnes, M-7 Author(s): Davis L. Tonks, X-4 Joint AIRAPT/APS Conference on High-Pressure Submitted to: Science and Technology, Colorado Springs, Co American Physical Society DISTRIBUTION OF THIS DOCUMENT IS UNLIMITED 10 We want the second second MASTE



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MEASUREMENT AND ANALYSIS OF THREE 1.5-GPA SHOCK-WAVE PROFILES IN COPPER

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Three wave-profile experiments were performed on OFE fully annealed (600 °C for one hour) copper us ing a 101.6-mm-diam gas gun at impact velocities of 86 m/s. A symmetric impact produced a 1.5 GPa shock vave in the target. A sapphire window was bonded to the front (non-impact) face of the target, and a four-detector push-pull velocity interferometer (VISAR) measured the velocity of the copper/sapphire interface. The impactor fluckness (4 mm) was the same in all experiments; the target thicknesses were 10, 20, and 30 mm. The stresses and strains, including the deviatorie stresses and strains, have been extracted to on these data using a quasi-tagrangian analysis. (The waves are not steady.) The use of three separate shots in Lagrangian analysis yields only approximate results for the deviatoric stresses; but the results for the normal stresses, and for the strains, are fairly accurate. Even though the strain rates fall in the Hopkinson bar regime, the mechanism of dislocation motion appears to be dislocation drag, as is the case for stronger shock waves in Cu.

INTRODUCTION

The three experiments presented here are part of a series of experiments to measure and model the elastic plastic response of OFE fully annealed copper subjected to shock and reshock and shock and release loading at initial pressures of 1.5 and 3 GPa. Some of the 3 GPa experiments and analyses have already been reported [1]. This paper presents the very lowest pressure experiments and analyses.

EXPERIMENTS.

A 101.6-mm illiam gas gun was used to produce a symmetric impact in OFE copper at an impact velocity of about 86 m/s. The copper in both the impactor and the target was fully an incaled (600°C for one hom). A single crystal sapphire window [2] was bounded to the front (non-impact) face of the copper target.

The impactor velocity was measured with an array of electrical contact plus recording on digitizers with a resolution of 5 us. The velocity of the larget/sapplific interface was measured with a four detector push pull VISAR [3] recording on digitizing oscilloscopes sampling at either 4 or 2 grassamples/s.



Figure 1. Experiment 111207; 10 num target. The velocity is at the Cu/sapphire interface.

In these three low stress, low strain rate experiments the impactor thickness was 4 mm, and the target thicknesses were 10 mm, 20 mm, and 30 mm. Details of the experiments are given in Tables 1 and 2. The wave profiles at the target/sapplific window interfaces are plotted in Figs. 1-3.

Fable 1. Dimensions of components in the experiments.

Vinitia	
Sapphire Window	
Ettann	
(num)	
50 11	
561.61	
561.0	

Table 2. Impact velocities and violation	Tal	ble	2.	Impact	velocities	and	VISAR	setu
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	Impactor		Fringe
Experiment	Velocity	Tilt	Constant
No.	(m/s)	(mrad)	(n\/s)
H1207	85.7 ±0.1	0.85	39.07
H1208	85.2 ±0.3	0.66	39.07
111230	86.2 ±0.1	1.21	39.07



Figure 2. Experiment H1208; 20-mm target. The velocity is at the Cu/sapphire interface.



Figure 3. Experiment 111230; 30 unit target. The velocity is at the Cu/supplifier interface.

THEORETICAL ANALYSIS

The particle velocity data were smoothed by local least squares. filling to eliminate local ringing that this ged the sign of the acceleration This resulted in an drastic changes to the data.

Theoretically, Lagrangian analysis requires progressive simplifies of the same wave. The

data available here, however, are from three different experiments with three different sample thicknesses. (See Table 1.) Shot-to-shot variations cloud the straightforward application of the technique to these data. For example, as seen in Figs. 1-3, experiment 11230 has a somewhat larger peak velocity than the others. In order to apply the Lagrangian method, these variations must be eliminated in some fashion. Therefore, the profiles for experiments H1207 and H1208 were scaled in velocity to make their peak velocities identical to that of shot H1230. (The adjustments were minor, less than eight percent.) The method of analysis used here consists of a Lagrangian analysis together with this and other methods of eliminating shot-to-shot variations (to be described shortly). It will be called a pseudo-Lagrangian analysis.

To estimate the extent of the error, each wave was analyzed using the steady wave analysis of Wallace [4] and the results were compared to that of the pseudo-Lagrangian method used here. The steady wave results are acty not correct, but provide some sort of bound on the correct result. As seen in Figs. 4–7, the results from both analyses compared fairly closely for all quantities except the deviatorie stress, which is the quantity most sensitive to error. Nevertheless, the two versions of the deviatorie stress, Fig. 5, are similar, differing at most by about 30%.



Figure 4. Calculated compression and plastic strains versus time for the 111208 gauge posttion.

A 3% elastic impedance correction was qualitied in the measured interface velocity to esti-

mate the in situ Cu particle velocity without the presence of the sapphire. For a correction this small, the elastic theory is adequate, [5].

The Lagrangian method of paths was used, [7]. Each profile was divided into three monotonic parts separated by the peak of the first rise, the following dlp, and the top of the plastic wave. Points with the same relative velocity in corresponding parts in the three profiles were connected to form paths.

The equations used for the pseudo-Lagrangtan analysis are

$$\sigma(t) = -\rho_{0} \int \left(\frac{\partial u}{\partial t}\right)_{h} dh \Big|_{t}$$
(1)

and

$$\varepsilon = \left. \int \left(\frac{\partial u}{\partial h} \right)_{t} dt \right|_{h}.$$
 (2)

where n is the particle velocity, σ is the normal stress defined posilive in compression, r is the compression or volumetric strain, h is the Lagrangian position, ρ_0 is the initial density, and t is the time. The normal stress and compression were found at the sample thickness (20 mm) corresponding to shot 111208 using these formulas. Eq. (1) was integrated along a time of constant time in the h t plane inward from an undis indeed position in front of the shock wave. Eq. (2) was integrated along a line of constant time in the h t plane inward from an undis indeed position in front of the shock wave. Eq. (2) was integrated along a line of constant h corresponding to the 111208 sample thickness. The 111208 position is in the center of the data and, times, extrapolations there are infinited.

Once it and covere obtained, the thermoelas (te equations of Wallace [4] were used to obtain the deviatorie stress and plastic strains. The material parameters used are given to [5,6].

The steady wave procedure used to obtato the steady wave results are also described to these references. In the calculations here, the else to the first peak was assumed elastic and the vebetty of point "c" used to the analysis [4] was taken to be 10 m/s.

The absolute thinking measurements of the transit time for the three profiles proved to be too inaccurate to obtain physically reasonable results; negative deviatoric stresses wice obtained. Using the points of first motion to add in aligning the flore records in frace also fed to negative deviatoric stresses. To set the relative thinking of the three profiles, the produced the first rise were taken to have to were the sample thicknesses at the longitudinal sound velocity. This assumption, together with the velocity versus time record for each profile, established the timing for all profile points.



Figure 5. Calculated deviateric stress versus time for the 11208 gauge position.



Figure 6. Calculated deviatode stress versus plastic stratu for the 111208 gauge position. The X is a calculated value for the internal for the strength at the top of the shock wave.

Doth Amean and quodratic Atting of the data were fitted Both gave about the same results for the stresses and stratus at the 414208 target thickness. The calculated peak deviatoric stress, however, was about 10% hower for the quadratic At than for the linear At. For the Utelenesses corresponding to the other two experiments the quadratic fit gave regions of negative deviatoric stress. These results are not shown here. Since the linear fit did not produce this problem, only the linear results are given here. One would think that the quadratic fit would give better results because it was based on all of the data. The reason it did not may be due to the shot to shot variations.

An additional restriction is that only data from shots H1208 and H1230 were used in the linear fit. Using data from shots H1207 and H1208 in the linear fit also produced negative deviatoric stresses.



Figure 7. Calculated plastle strain rate ver sus plastic strain for the 111208 gauge position.

We note that the devialance stress used here is: $\mathbf{r} = \begin{pmatrix} -\frac{1}{2} \end{pmatrix} (\mathbf{r}_{x1} - \mathbf{r}_{yy})$, where \mathbf{r}_{y1} and \mathbf{r}_{y1} are the stresses along, and normal to, the shock direction, respectively.

The curves of the deviatoric stress versus plastic strain, Fig. 6, and of the shain rate versus plastic strain, Fig. 7, closely reachine those hurstronger ('n shock wave profiles that are dominated by dislocation drag [1]. The very last rise in plastic strain rate with plastic strain and the peaking of the deviatorie stress about unit way in plastic strain are typical of shock wave dislocation drag processes in which the tuilital stress loading is higher than any initial barders and laster than any buildup of internal work hardening. Hence, the mechanism of plastic tlaw here is probably infinenced by dislocation drag about

Evidence for this was obtained by calculating, for the end of the shock path of Fig. 6, the me chantcal threshold stress using the model of [8]. Decause of the small strains involved, the results of this model should be considered fairly rough. accurate to, say, 40%. The resulting value of 20 MPa for the mechanical threshold stress should be compared with the final deviatoric stress of Fig. 6. Due to the uncertaintles involved, these two values are roughly the same. As can be seen, after the first rise the deviatorie stresses of the shock path lie above this value, which is an upp**e**r bound for the evolving threshold stress, indicating that the dislocations are being driven by stresses greater than the opposition of internal barriers. Hence, even though the strain rates here fall in the range of Hopkinson bar data, which is dominated by thermally activated dislocation motion, the stress loading here is litgher than internal barriers and so abrupt that dislocation drag is the mechanism of dislocation motion.

ACKNOWLEDGMENTS

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REFERENCES

- [1] R. H. Warnes and D. L. Tonks, "Measure ment and Analysis of 3 GPa Shock Wave Profiles in Annealed OFE Copper," in Proceedings of the APS Topical Conference on Shock Compression of Condensed Matter, 1989, pp. 329–332.
- [2] E. M. Burker and R. E. Hollenbach, J. Appl. Phys.44, pp. 4208 (1226) [1970).
- [3] W. F. Hemsing, Rev. Sci. Instrum. 50(11, pp. 73–78 (1979).
- [4] D. C. Wallace, Phys Rev. B 22, op 1477 1486 and 1487 1494 (1980).
- [5] D. L. Fouks, J. Appl. Phys. 66, pp. 1954–1960 (1989).
- [6] D. L. Tauks, "The DataShoP, A Database of Weak Shock Constitutive Data," Los Alames National Eaboratory Report LA (2068) MS, (1994).
- [7] L. Seaman, J. Appl. Phys. 45, p. 4303 (1974).
- [8] J. N. Julmson and D. L. Touks, "Dynamic Plasticity in Transition from Thermal Activation to Viscous Drag," in *Proceedings of the* APS Topfeul Conference on Shock Compression of Conference on Shock Compression of Conference Mutter, 1991, pp. 374– 378.

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