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# A Numerical Model Study of Burning and Detonation in Small PETN-Loaded Assemblies

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## A NUMERICAL MODEL STUDY OF BURNING AND DETONATION IN SMALL PETN-LOADED ASSEMBLIES

by

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#### ABSTRACT

A simple model has been used in the 2DL code to calculate the burning and detonation in Robert Dinegar's all-PETN hot-wire assemblies. The model includes (1) a constant velocity ignition front, (2) conductive  $\dot{\mathbf{X}} = \mathbf{BP}^n$  burn of particles behind the ignition front, (3) inertial confinement, and (4) shock initiation of detonation in the assembly transition tube PETN charge. The surface-to-volume ratio of the burning particles and the ignition front velocity were varied in a parameter study of the model.

#### I. INTRODUCTION

An explosive that is initially burning may sometimes detonate; the behavior change is called the deflagration-to-detonation transition (DDT). The term is nonspecific and is used in a variety of circumstances occurring in explosive gases, liquids, solids, and mixtures of these phases. The DDT is considered contributory in some explosive accidents. However, DDT may also be used in design applications of explosive devices.

This report describes a model study for the deflagration-to-detonation transition in pressed secondary explosive. In particular, the study is restricted to one of the hot-wire-ignited, PETN-loaded experimental devices studied by Robert Dinegar.<sup>1</sup>

#### **II. THE EXPERIMENTAL ASSEMBLY**

The PETN-loaded, hot-wire-ignited, SE-1 assembly is shown in Fig. 1. This device uses an SE-1 detonator mechanical assembly, which includes the plastic head with mounted bridgewire and the brass tube, interior to a steel confining jacket. The SE-1 is loaded partially with pressed PETN, the donor charge. Atop the donor charge is a steel tube loaded with PETN pressed to a lower density, the transition charge. The steel transition tube fits inside the brass SE-1 tube and confines the donor charge. The reduced inside diameter of the transition





Hot-wire-ignited DDT assembly (Dinegar).

tube is necessary to the functioning of the device. The assembly is fired by heating electrically, but not exploding, the bridgewire.

Dinegar varied the pressing densities. The donor charge was pressed to 1.40 and 1.60 g/cm<sup>3</sup>. The transition charge was pressed to densities of 0.8, 1.00, 1.20, 1.40, and 1.60 g/cm<sup>3</sup>. Successful detonations were produced with donor charges pressed to 1.40 and 1.60 g/cm<sup>3</sup> and with transition charges pressed to 0.8, 1.00, and 1.20 g/cm<sup>3</sup>, using coarse PETN and a 1.28-cm length.

From these successful systems, we have picked the 1.60-g/cm<sup>3</sup> donor charge and 1.20-g/cm<sup>3</sup> transition charge system for the model study.

#### **III. THE MODEL**

The computer model simulates the deflagration-to-detonation transition with a sequence of events beginning with the donor charge prepressurized and with a small region of donor explosive ignited near the hot wire. An ignition front spreads outward at constant velocity with explosive burning behind the ignition front. Then as the donor charge burns, pressure builds in the donor charge and a pressure wave is sent into the PETN transition charge. This pressure wave compacts some of the transition charge and forms a shock, which grows into a detonation wave traveling down the transition tube. No two-phase flow is allowed; that is, the gaseous products do not flow relative to the solids from which they arose. The ignition front is not calculated from convection processes but is imposed as a constant velocity wave, of which the velocity is a model parameter.

Calculations herein were done with the 2DL code,<sup>2</sup> a two-dimensional, Lagrangian, finite difference, fluid dynamics computer code, using cylindrically symmetric (r,z) geometry. The initial problem SE-setup is shown in Fig. 2; note that only the interior SE-1 detonator assembly with transition tube is included. The steel case and the air gap between the SE-1 and the steel case were omitted because preliminary calculations indicated that detonation or failure to detonate resulted before the brass expanded to close the air gap. Those calculations seem to indicate that the steel case holds the assembly together during the initial stages of pressure buildup in the donor charge, keeping the transition tube from popping out, but not radially confining the assembly. The heavy lines in the setup figure, with one exception, are material bound-



SE-1 hot-wire assembly, 2DL problem setup, cylindrical (r,z) geometry.

aries. The exception is the vertical line parallel to the cylindrical axis in the left of the figure. This line is a slip surface that continues throughout the mesh length. The slip surface was included to allow compaction of the low-density PETN ( $\rho_0 = 1.20 \text{ g/cm}^3$ ) in the transition tube. Code restrictions require that a slip surface continue through the entire mesh. The cell mesh is not attached across the slip surface; that is, the inner mesh may move independently of the outer mesh. For this problem set there are no frictional forces at the slip surface.

Dimensions and the numbers of finite difference cells are listed in Table I and shown in Fig. 2 for the initial setup regions. Materials occurring on both sides of the slip surface, for instance, the plastic and the 1.60-g/cm<sup>3</sup> PETN, have identical material constants in each region and act as a single material.

In the donor charge burning is allowed and simulated by the simple porous-bed burn model,<sup>3</sup> in which the explosive is assumed to be burning on particle surfaces according to the burn surface regression rate law of  $\dot{X} =$ BP<sup>n</sup> (cm/µs) after the passage of an ignition front. The ignition front spreads from already ignited material at a

#### **INITIAL SETUP**

Material	Inner Radius (cm)	Outer Radius (cm)	NR Number of Cells	Length (cm)	NZ Number of Cells
	(Inside S	Slip Surfa	ce)		
Plastic	0.0	0.1025	5	0.60	30
PETN, $\rho_0 = 1.60 \text{ g/cm}^3$ Ignited	0.0	0.1025	5	0.02	1
PETN, $\rho_0 = 1.60 \text{ g/cm}^3$ Not ignited	0.0	0.1025	5	0.62	31
PETN, $\rho_0 = 1.20 \text{ g/cm}^3$ Forest Fire	0.0	0.1025	5	0.96	48
	(Outside	Slip Surfa	ice)		
Plastic	0.1025	0.38	14	0.60	30
PETN, $\rho_0 = 1.60 \text{ g/cm}^3$ Not ignited	0.1025	0.38	14	0.64	32
Steel	0.1025	0.38	14	0.96	48
Brass	0.38	0.50	12	2.20	110

constant velocity. Explosive decomposition is described by the equation

$$\frac{\mathrm{d}\mathbf{W}}{\mathrm{d}t} = -\left(\frac{\mathbf{S}}{\mathbf{V}}\right)_{\mathbf{0}} \mathbf{W}^{\mathbf{q}} \mathbf{B} \mathbf{P}^{\mathbf{n}}$$

where

W = mass fraction of solid explosive,  $(S/V)_0 = initial surface-to-volume ratio (1/cm),$ q = particle-geometry-related exponent, andP = pressure (Mbar).

The exponent q (in the  $W^q$  term) has values of 2/3 for spherical, 1/2 for cylindrical, and 0 for planar particles. The values<sup>4</sup> of B, n, and q are listed in Table II. The values of  $(S/V)_0$  and the ignition velocity were varied in the model parameter study. Initial 0.4-kbar gas pressure is set throughout the donor charge by setting the initial solid-PETN mass fraction there to 0.998. Experimentally, ignition is started with one atmosphere donor charge pressure and pressure increases over a relatively long time. Dinegar has observed delays of a few milliseconds in the functioning of the assemblies. Also, because the mass consumption rate is proportional to pressure and because pressure should be approximately proportional to the mass fraction of gas products, we expect that the initial pressure growth in time is exponential.

Shock initiation of detonation is allowed in the transition tube and is modeled by the Forest Fire<sup>3,5</sup> shock-induced reaction rate model, a one-step solid-to-products rate model calculated from the equation of state and experimental shock Hugoniot and Pop plot for the explosive. [The Pop plot is the graph of distance to detonation (run) versus initial shock pressure, as measured in the wedge test, and is named for Alphonse Popolato.<sup>6</sup>] Briefly, the Forest Fire rate is the explosive decomposition rate necessary to accelerate a shock along the time-distance-state space line determined by the Pop plot and Hugoniot, using the Pop plot as a shock growth curve. In the calculations presented here, the rate function is given as a function of local pressure and is used throughout the transition charge; rate function constants are listed in Table II.

The HOM<sup>7</sup> equation of state is used throughout this calculation. This equation of state represents solids by a Grüneisen expansion off the shock Hugoniot with temperatures calculated by the Walsh and Christian<sup>8</sup> technique. The gas products are represented by a beta

#### TABLE II 2DL INPUT CONSTANTS

Porous-bed burn constants for PETN, where  $\rho_0 = 1.60 \text{ g/cm}^3$ .

$$\frac{dW}{dt} = - \left(\frac{S}{V}\right)_0 W^q B P^n \quad \left(\frac{1}{\mu s}\right) \ , \label{eq:dw}$$

where q = 0.667, B = 0.0119, and n = 1.0, with P = Mbar. HOM CONSTANTS - PETN, RHO = +1.20

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+0.01272 +2.3 +0. +0.
+0. +7.35770705112E+00+2.39083797487E+01+1.16201137629E+02
+1.96236668634E+02+1.16157063364E+02+1.15 +0.24
+8.33333333333333E-01+0.400232 +0. +0.
+0.01272
                                     +0.
+0.
                                              +0.
+330.430499777 +0.000001
                                                                     +0.
                                              +0.
+0.
                       +0.
-3.17802221190E+00-2.35331643478E+00+2.17043373808E-01+1.17479938619E-02
-5.94101012630E-03-1.42326845168E+00+4.77286576745E-01+6.03149941862E-02+4.05801307902E-03+1.08173212997E-04+7.93913046699E+00-4.31173734538E-01
+1.03363956902E-01-2.32720523000E-02+2.12830586356E-03+0.5
+0.1
FOREST FIRE POLYNOMIAL COEFFICIENTS
 15 PETN, RHO=1.20, ESTIMATED POP PLDT AND HUGON10T, BKW GAS
1.44000000000E-01 2.0000000000E-03 DW/DT=-W#F(P)
-1.22090225473E+18 1.37845714582E+18-6.99723707168E+17 2.10990512254E+17
-4.20660384487E+16 5.84118991286E+15-5.79662003909E+14 4.15224044246E+13
-2.14323058755E+12 7.87550287132E+10-2.01421458481E+09 3.46783232499E+07
-3.86339277739E+05 2.80914048267E+03-1.23187022045E+01
HOM CONSTANTS - STEEL, RHD = +7.917
                                        +000+0.
+4.58
                -001+1.51
                                                                     +0.
+++56 --5.82582587453+003-7.03211954024+003-4.82670213894+003
-1.46678402118+003-1.66591615983+002+2.02 +1.07 -001
+1.26310471138-001+1.17 -005+0. +0.
+0.
                                   -005+0.
+1.26310471138-001+1.17
             +0.000001
                                          +0.
+300.
                                                                     +0.
+0.
                       +0.
                                             +0.
```

HOM CONSTANTS - BRASS, RHO = +8.413

HUM CUNSTANTS - PLASTIC, RHO = +1.18

+0.2432	+000+1.5785	+000+1.0	-004+0.0	+000
+0.0	+000+5.293802	43506+000-4.249503	371368+000-1.5505552	/6332+001
-3.086380755	72+001-1.467081	93739+001+1.0	+000+0.35	+000
+8.47457627	-001+1.0	-004+0.	+0.	
+3.0	+002+1.0	-086+0.J	+U.D	
+0.	+0.	+0.		

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HOM CONSTANTS - PETN, RHO = +1.60
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FOREST FIRE POLYNOMIAL COEFFICIENTS

13 PETN, RHO=1.60, PCJ=0.2564, OCJ=0.7818, VCJ=0.46118 2.56326000000E-01+0.003 OW/OT=-W $\approx$ F(P) -1.45372719700E+13 2.13878630671E+13-1.39169518661E+13 5.27427850939E+12 -1.29042984524E+12 2.13542626252E+11-2.43534878228E+10 1.91547535104E+09 -1.02479283910E+08 3.63112108861E+06-8.24093189751E+04 1.22282347454E+03 -1.08522971453E+01 expansion off the BKW<sup>9,10</sup> calculated detonation isentrope. Solid and gas product mixtures are calculated by simultaneous solution of the solid and gas equations for pressure and temperature equilibrium, assuming ideal volume and energy partition according to mass fraction. The HOM constants for each material are listed in Table II.

Because Forest Fire depends upon the shock Hugoniot and Pop plot, and HOM depends upon the shock Hugoniot for the solid equation of state, these relations must be known for each material in the model to which they apply. Experimental data are available except for PETN with density  $\rho_0 = 1.20$  g/cm<sup>3</sup>, where

the Pop plot and Hugoniot need to be estimated. These estimates are made and the method for each is presented below. Experimental wedge data on PETN at other densities<sup>11-13</sup> are used heavily.

The Pop plot for  $\rho_0 = 1.20 \text{ g/cm}^3 \text{ PETN}$  is estimated by a line parallel to the  $\rho_0 = 1.40 \text{ g/cm}^3 \text{ PETN}$  Pop plot. We found this approach reasonable by inspection of the various PETN Pop plots in Fig. 3. [The least-square linear ln (run) = a + b ln (P) fits are listed in Table III.] By paralleling the  $\rho_0 = 1.40 \text{ g/cm}^3$  line, it is sufficient to estimate only one point on the line. Again looking at the Pop plot in Fig. 3, a 0.3-cm distance to detonation lies interior to the data range for each of the data sets and



Fig. 3. Distance to detonation as a shock pressure function at various densities,  $(g/cm^3)$ . The 1.20- $g/cm^3$  line is estimated from the wedge test data.

#### TABLE III

## PETN POP PLOTS— LEAST-SQUARE FITS FROM WEDGE DATA

ln(run) = a + b ln (P), where

run = distance to detonation (cm), and

P = initial shock pressure (Mbar).

 $(P^* = \text{shock pressure for run} = 0.3 \text{ cm.})$ 

Initial Density,			Shock Pressure,		
ρο	Const	tants	P*		
(g/cm <sup>3</sup> )	a	b	(Mbar)		
1.00	-16.76803	-2.62043	0.00263		
1.40	-12.42600	-2.38626	0.00907		
1.60	-9.44241	-1.94113	0.01435		
1.72	-7.10742	-1.55638	0.02253		
1.75	-8.64881	-2.00592	0.2444		
	Estimated	Pop Plot			
1.20	-13.78	-2.40	0.0053		

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serves as the distance to detonation for the unknown point, with the corresponding pressure to be determined.

The pressure is determined from a graph of the pressure required for a 0.3-cm run from each of the Pop plot lines as a density function in log-log coordinates in Fig. 4. The points define the solid line at  $\rho_0 = 1.40$  and 1.60 g/cm<sup>3</sup>, which gives a shock pressure of 0.053 Mbar for  $\rho_0 = 1.20$  g/cm<sup>3</sup>. Note that the point for  $\rho_0 = 1.00$  g/cm<sup>3</sup> lies nearly on the line. The Pop plot so estimated is then *l*n (run) = -13.78 - 2.40 ln (P) with run = cm and P = Mbar.

The shock Hugoniot line  $(U_s = C + SU_p)$  is estimated from a uniform Hugoniot representation for each of the densities by fitting each of the  $(U_p, U_s)$  data sets to a straight line with slope 2.3, letting the intercept at  $U_p = 0$ fall where it may (see Fig. 5 and Table IV). These intercepts are plotted as a density function; in particular, the intercepts are graphed against  $\rho_{max}/\rho_0$  in log-log coordinates in Fig. 6. The motivation for this method came from an observation on the PETN data at the extremes of the density range. First, Olinger and Cady<sup>14</sup> gave a Hugoniot for 1.774-g/cm<sup>3</sup> PETN derived from x-ray diffraction patterns taken at pressures up to 0.100 Mbar. Their quadratic Hugoniot relation is well approx-



Fig. 5.

PETN shock Hugoniots. Data points are from Los Alamos group WX-2 wedge tests. Lines are least-squares fits with constant slope of 2.3. Labels on each line indicate the density  $(g/cm^3)$ .

#### TABLE IV

## SHOCK HUGONIOT INTERCEPTS FOR CONSTANT SLOPE $U_s = C_0 + 2.3 U_p$ LINES

Initial Density,			
ρο		C	Fit C <sub>0</sub>
(g/cm <sup>3</sup> )	$\rho_{max}/\rho_0$	( <b>cm</b> /µs)	(cm/µs)
1.75	1.01600	0.2469	0.2526
1.72	1.03372	0.2254	0.2254
1.60	1.11125	0.1422	0.1371
1.40	1.27000	0.0488	0.0495
1.00	1.778	0.0033	0.0019

imated by the linear relation  $U_s = 0.232 + 2.306 U_p$  (cm/µs). Secondly, the data Seay and Seely<sup>12</sup> gave for 1.00 g/cm<sup>3</sup> PETN is well fit by the line  $U_s = 2.3 U_p$ , if the shock velocities are corrected by a factor of 0.9. (The velocities were estimated to be 10% too high because the velocities reported were average velocities over the entire run to detonation and not initial velocities.)



Fig. 4.

Shock pressure needed for a 0.3-cm distance to detonation as a function of PETN density. The solid line is drawn through the 1.40 and 1.60 g/cm<sup>3</sup> points.



Fig. 6. PETN constant slope Hugoniot intercepts as a density function. Labels at the points indicate density  $(g/cm^3)$ .

The least-squares fit of the intercepts as a density function is  $C = \beta_1 \exp(-\beta_2 \rho_{max}/\rho_0)$ , where  $\rho_{max} = 1.778$ g/cm<sup>3</sup>,  $\beta_1 = 171.60519$  (cm/µs), and  $\beta_2 = 6.41846$ , with a coefficient determination  $r^2 = 0.9988$ . Evaluation of the fit at  $\rho_0 = 1.20$  g/cm<sup>3</sup> gives the Hugoniot relation U<sub>s</sub> = 0.0127 + 2.3 U<sub>p</sub>. Note also that evaluation at  $\rho_0 =$ 1.00 g/cm<sup>3</sup> gives an intercept of C<sub>0</sub> = 0.0033 cm/µs.

The Forest Fire rate calculation based on these estimates is listed in Table V.

#### **IV. CALCULATION SETS**

For each of the three ignition velocities—0.05, 0.10, and 0.20 cm/ $\mu$ s— a calculation was made for various surface-to-volume ratios in the porous-bed burn model. Graphic 2DL output is shown in Figs. 7a-7d, 8a-8f, and 9a-9d, with the ignition velocity held constant in each set. The ignition front position is independent of the surface-to-volume ratio, is thus the same in time for each set calculation, and so is plotted only once for each set. Sequential pressure contours are presented for each calculation with mass fraction contours included for those times showing reaction progress. Table VI lists the figure numbers.

At the bottom of each pressure contour frame are listed the time ( $\mu$ s) from the beginning of the calculation, the pressure contour interval ( $\Delta P = kbar$ ), and the maximum pressure ( $P_{max} = kbar$ ) over all cells. At the bottom of the mass fraction plots are listed the time and mass fraction contour interval ( $\Delta W$ ), and at the bottom of the ignition front position plot is listed the time only.

In examining the pressure plot sequences, the pressure contour interval and the maximum pressure should be noted carefully. The contour interval is varied to limit the number of contour lines so a rapid increase in pressure (say a doubling) from one frame to the next may not be apparent from contour lines because the interval may also increase and the number of lines would stay about constant.

#### **V. CONCLUSIONS**

Figure 10 shows general features of the model calculations as a graph of the maximum pressure taken from the entire mesh plotted as a function of ignition time. The calculation set with 0.05-cm/µs ignition velocity shows steady pressure increase until 12 µs, followed by a decrease in pressure. The maximum in the curve occurs soon after the arrival of the ignition front at the top of the donor charge, see Fig. 7a. Apparently, the reason for the pressure drop is that the brass case expands faster than gas is evolved to fill the increased volume; the long time required to get the entire mass burning allows for greater acceleration of the brass, see Figs. 7b and 7d. The calculation set with 0.1-cm/µs ignition velocity shows a maximum in the pressure-time curve in Fig. 10 for surface-to-volume ratios of 400 and 500/cm, but steady pressure increase for (S/V) of 600 and 800/cm. The line for (S/V) = 500/cm appears to decrease but surely must induce shock initiation into the PETN transition charge because the pressure is high enough. Again, the drop is due to case expansion before shock-induced decomposition in the transition charge has proceeded very far. The calculation set for 0.2-cm/µs ignition velocity shows similar behavior with shorter times. The overall behavior is simply stated—the transition to detonation is achieved most rapidly where a high mass burn rate occurs in compacted explosive. Slow ignition delays the gas production and allows greater expansion, and thus gives lower density in the burning bed.

## FOREST FIRE CALCULATION FOR PETN $\rho_0 = 1.20 \text{ g/cm}^3$

INPUT

PETN. RHO=1.20. ESTIMATED POP PLOT AND HUGONIOT. BKW GAS 28.JAN81 RHO = 1.20000POP PLOT, LN(RUN) = A1 + A2 H LN(P-A3), A1 = -1.378000E+01 A2 = -2.400000E+00 A3 = 0. REACTION HUGONIOT, US = C + SHUP, C = 1.300000E-02 S = 2.400000E+00 CJ DETONATION PRESSURE = 1.440000E-01HOM EQUATION OF STATE CONSTANTS PETN, RHO=1.20, US=C+2.3"UP UNREACTED EXPLOSIVE 1.2720000000E-02 2.300000000E+00 0. Ο. Ο. 7.35770705112E+00 2.39083797487E+01 1.16201137629E+02 1.96236668634E+02 1.16157063364E+02 1.1500000000E+00 2.4000000000E-01 8.33333333333E-01 2.3200000000E-04 0. Ο. 3.30430499777E+02 1.0000000000E-06 0. Ο. ٥. Ο. Ο. DETONATION PRODUCTS -3.17802221190E+00-2.35331643478E+00 2.17043373808E-01 1.17479938619E-02 -5.94101012630E-03-1.42326845168E+00 4.77286576745E-01 6.03149941862E-02 4.05801307902E-03 1.08173212997E-04 7.93913046699E+00-4.31173734538E-01 1.03363956902E-01-2.32720523000E-02 2.12830586356E-03 5.00000000000E-01 1.00000000000E-01

## RATE CALCULATION

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PETN, RHO=1.20, POP PLUT, LN(R	ESTIMATED P UN) = A1 + A	POP PLOT AND 2"LN(P-A3),	HUGONIOT, BK AI = -1.378	(W GAS 8000E+01 A2	28JAN81 = -2.400000E	RHO = 1.2000 +00 A3 = 0.	0	
REACTION HUGON	110T, $US = C$	: + S <sup>#</sup> UP,	C = 1.300	000E-02 S	= 2.400000E	+00		
RUN	P	v	UP	US	W	RATE	TEMPERATURE	TIME
3.11136	.00200	.55052	.02378	.07008	00709	3 29215 04	1040 24622	
1.17579	.00300	.53970	.02968	. 08423	• 5 57 50 QQ704	J. 2021C-04	1940./40/3	37.30366
. 58949	.00400	. 53304	.03466	.00618	00610	1.42010-03	210/.58/62	11.69599
.34506	.00500	52841	.03905	10671	00561	4.20/2C-UJ	2552.02805	5.12700
.22277	.00600	52494	.04302	11624	00470	9.3702C-03 1 90005 00	2509.09250	2.70183
.15388	.00700	.52222	.04667	12500		1.04000-02	2045./0109	1.60001
.11169	.00800	.52001	.05007	13316	00741	5 07/05 02	2/03.32529	1.02/08
.08419	.00900	.51817	.05326	14082	00281	J.U/40E-UZ 7 67115 00	20/1./3009	.69942
.06538	.01000	.51660	05628	14807	00226	1 10035 01	290/.30/9/	.49829
.01239	.02000	.50796	.08067	. 20661	08723	1 26285-01	JUJJ.0JJY/ 3607 00361	. 30/88
.00468	.03000	.50406	. 09939	25154	08275	5 35605.00	3027.22341	.04985
.00235	.04000	.50171	.11517	28042	07834	1 53105+00	5945.00//2 h159 h1176	.01540
.00137	.05000	.50009	. 12908	32270	07384	1.55102+01	4172.411/4	.000/5
.00089	.06000	49890	14165	36207	06015	7 10765.01	4299.00333	.00333
00061	07000	49090	16700	• 222 37	.90913	/.10/02+01	4407.52541	.00208
	.07000		16709	. 300/2	.9041/	1.30556+02	4490.09934	.00133
.00044	.08000	•49/21	.10398	.40055	.95884	2.2406E+02	4555.89/92	.00091
.00034	.09000	.49059	.1/409	.43081	.95305	3.6660E+02	4603.45182	.00064
.00026	.10000	.49606	-18365	.453/6	.94673	5.7395E+02	4641.85274	.00048
.00021	.11000	.49560	.19274	.47559	.93974	8.7958E+02	4671.13319	.00036
.00017	.12000	.49520	.20143	.49644	.93195	1.3136E+03	4692.81022	.00028
.00014	.13000	.49485	.20977	.51644	.92319	1.9332E+03	4707.90640	.00022
.00012	.14000	.49454	.21779	.53569	.91323	2.7897E+03	4717.23571	.00018
.00010	.15000	.49426	. 22553	.55426	.90177	4.0289E+03	4721.21584	.00015
.00008	.16000	.49400	.23301	.57222	.88839	5.7604E+03	4720.16041	.00012

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## **POLYNOMIAL FIT**

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PETN, RH0=1.20, ESTIMATED POP PLOT AND HUGONIOT, BKW GAS 28JAN81 RH0 = 1.20000 LN(RATE) = C(1) + C(2)#P + ... + C(M+1)#(P##M) C(1=1,15) = -1.2318702205E+01 2.8091404827E+03 -3.8633927774E+05 3.4678323250E+07 -2.0142145848E+09 7.8755028713E+10 -2.1432305875E+12 4.1522404425E+13 -5.7966200391E+14 5.8411899129E+15 -4.2066038449E+16 2.1099051225E+17 -6.9972370717E+17 1.3784571458E+18 -1.2209022547E+18

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PRESSURE	RATE	FIT	REL. ERROR
2.000000E-03	3.282067E-04	3.360700E-04	023959
3.000000E-03	1.428149E-03	1.391311E-03	.025794
4.000000E-03	4.287172E-03	4.139068E-03	.034546
5.000000E-03	9.576162E-03	9.658418E-03	008590
6.000000E-03	1.839967E-02	1.889484E-02'	026911
7.000000E-03	3.172286E-02	3.257393E-02	026828
8.000000E-03	5.074789E-02	5.133875E-02	011643
9.000000E-03	7.671084E-02	7.596307E-02	.009748
1.000000E-02	1.109302E-01	1.075202E-01	.030740
2.000000E-02	1.262566E+00	1.270871E+00	006578
3.000000E-02	5.355952E+00	5.374420E+00	003448
4.000000F-02	1.531000E+01	1.510351E+01	.013487
5.000000E-02	3.527619E+01	3.586095E+01	016577
6.000000E-02	7.107579E+01	7.059710E+01	.006735
7.000000E-02	1.305465E+02	1.294960E+02	.008047
8.000000E-02	2.240611E+02	2.271405E+02	013744
9.000000E-02	3.665977E+02	3.646305E+02	.005366
1.000000E-01	5.739454E+02	5.700362E+02	.006811
1.100000F-01	8.795843E+02	8.902111E+02	012082
1.200000F-01	1.313577E+03	1.301449E+03	.009233
1.300000E-01	1.933248E+03	1.941716E+03	004380
1.400000F-01	2.789713E+03	2.786098E+03	.001296
1.500000E-01	4.028898E+03	4.029805E+03	000225
1.600000E-01	5.760421E+03	5.760335E+03	.000015



Fig. 7a. lgnition front position. lgnition velocity = 0.05 cm/ $\mu$ s. Burning occurs behind the ignition front.



Fig. 7b. Pressure contours. Ignition velocity = 0.05 cm/ $\mu s$  and (S/V) $_0$  = 400/cm.  $\Delta P$  = contour interval (kbar).

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Flg. 7c. Pressure contours. Ignition velocity = 0.05 cm/ $\mu$ s and (S/V)<sub>0</sub> = 500/cm.  $\Delta P$  = contour interval (kbar).  $P_{max} = maximum pressure (kbar).$ 



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Fig. 7c. (cont) lgnition velocity = 0.05 cm/ $\mu$ s and (S/V)<sub>0</sub> = 500/cm.  $\Delta W$  = contour interval fraction.  $\Delta P$  = contour interval (kbar). P<sub>max</sub> = maximum pressure (kbar).



Fig. 7d. Pressure contours. Ignition velocity = 0.05 cm/ $\mu$ s and (S/V)<sub>0</sub> = 600/cm.  $\Delta P$  = contour interval (kbar). P<sub>max</sub> = maximum pressure (kbar).



Fig. 7d. (cont) lgnition velocity = 0.05 cm/ $\mu$ s and (S/V)<sub>0</sub> = 600/cm.  $\Delta W$  = contour interval fraction.  $\Delta P$  = contour interval (kbar). P<sub>max</sub> = maximum pressure (kbar).



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Fig. 8a. lgnition front position. lgnition velocity = 0.10 cm/ $\mu$ s. Burning occurs behind the ignition front.



Fig. 8b. Pressure contours. Ignition velocity = 0.10 cm/µs and  $(S/V)_0 = 200/cm$ .  $\Delta P = contour interval (kbar)$ .



Pressure contours. Ignition velocity = 0.10 cm/ $\mu$ s and (S/V)<sub>0</sub> = 400/cm.  $\Delta P$  = contour interval (kbar). P<sub>max</sub> = maximum pressure (kbar).



Fig. 8c. (cont) Mass fraction (W) contours. Ignition velocity = 0.10 cm/ $\mu$ s and (S/V)<sub>0</sub> = 400/cm.  $\Delta$ W = contour interval fraction.



Pressure contours. Ignition velocity = 0.10 cm/ $\mu$ s and (S/V)<sub>0</sub> = 500/cm.  $\Delta P$  = contour interval (kbar). P<sub>max</sub> = maximum pressure (kbar).



Fig. 8d. (cont) lgnition velocity = 0.10 cm/ $\mu$ s and (S/V)<sub>0</sub> = 500/cm.  $\Delta P$  = contour interval (kbar).  $\Delta W$  = contour interval fraction. P<sub>max</sub> = maximum pressure (kbar).



Fig. 8e. Pressure contours. Ignition velocity = 0.10 cm/ $\mu$ s and (S/V)<sub>0</sub> = 600/cm.  $\Delta P$  = contour interval (kbar). P<sub>max</sub> = maximum pressure (kbar).



Fig. 8e. (cont) lgnition velocity = 0.10 cm/ $\mu$ s and (S/V)<sub>0</sub> = 600/cm.  $\Delta P$  = contour interval (kbar).  $\Delta W$  = contour interval fraction.  $P_{max}$  = maximum pressure (kbar).



Fig. 8f. Pressure contours. Ignition velocity = 0.10 cm/ $\mu$ s and (S/V)<sub>0</sub> = 800/cm.  $\Delta P$  = contour interval (kbar). P<sub>max</sub> = maximum pressure (kbar).



Fig. 8f. (cont) Mass fraction (W) contours. Ignition velocity = 0.10 cm/ $\mu$ s and (S/V)<sub>0</sub> = 800/cm.  $\Delta$ W = contour interval fraction.

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Fig. 9b. Pressure contours. Ignition velocity = 0.20 cm/ $\mu$ s and (S/V)<sub>0</sub> = 400/cm.  $\Delta P$  = contour interval (kbar). P<sub>max</sub> = maximum pressure (kbar).



Fig. 9b. (cont) lgnition velocity = 0.20 cm/ $\mu$ s and (S/V)<sub>0</sub> = 400/cm.  $\Delta P$  = contour interval (kbar).  $\Delta W$  = contour interval fraction.  $P_{max}$  = maximum pressure (kbar).



Pressure contours. Ignition velocity = 0.20 cm/ $\mu$ s and (S/V)<sub>0</sub> = 500/cm.  $\Delta P$  = contour interval (kbar).  $P_{max}$  = maximum pressure (kbar).

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Fig. 9c. (cont) Ignition velocity = 0.20 cm/ $\mu$ s and (S/V)<sub>0</sub> = 500/cm.  $\Delta P$  = contour interval (kbar).  $\Delta W$  = contour interval fraction.  $P_{max} = maximum pressure (kbar).$ 



Fig. 9d. Pressure contours. Ignition velocity = 0.20 cm/ $\mu$ s and (S/V)<sub>0</sub> = 600/cm.  $\Delta P$  = contour interval (kbar).  $P_{max} = maximum pressure (kbar).$ 



Fig. 9d. (cont) Mass fraction (W) contours. Ignition velocity = 0.20 cm/ $\mu$ s and (S/V)<sub>0</sub> = 600/cm.  $\Delta$ W = contour interval fraction.

## TABLE VI

Ignition Front Plot (Figure No.)	t Pressure-Mass Fraction P (Figure No.)					
		(S/V)	Ratio (	1/cm)		
	200	400	500	600	800	
7a		<u>7b</u>	7c	7d		
8a	<b>8</b> b	8c	8d	8e	8f	
9a		9Ь	9c	9d		
	Ignition Front Plot (Figure No.) 7a 8a 9a	Ignition Front Plot Press (Figure No.) 200 7a 8a 8a 9a	Ignition Front Plot (Figure No.)Pressure-Ma (Figure Na (S/V)2004007a7b8a8b8a8b9a9b	Ignition Front Plot (Figure No.)Pressure-Mass Fra (Figure N(Figure No.)(S/V) Ratio ( 2002004005007a7b7a7b7a7b8a8b8b8c9a9b	Ignition Front Plot (Figure No.)         Pressure-Mass Fraction F (Figure No.)           (S/V) Ratio (1/cm)           200         400         500         600           7a         7b         7c         7d           8a         8b         8c         8d         8e           9a         9b         9c         9d	

## CALCULATION SETS—FIGURE NUMBERS

<sup>s</sup>Figure numbers in boxes ( $\Box$ ) show no prompt detonation and a decreasing maximum pressure at late times in the calculation.

The steel transition tube effect can be seen especially in Figs. 8c and 8d, where there is a high-pressure annulus directly below the steel, just as the ignition front reaches the end of the donor charge. The center transition charge is compressed and relieves the pressure below it in the donor charge. At later times, case expansion is significant and drops the pressure radially, giving a maximum pressure at the center of the donor charge. The ignition velocity is expected to be less than the sound speed; in 1.70-g/cm<sup>3</sup> PETN the sound speed is about 0.14 cm/µs. The ignition velocities of 0.05 and 0.10 cm/µs may be physically possible and the 0.20 cm/µs may not be, but it was included to display the model sensitivity to this variable. Because no ignition mechanism is in the model and because ignition may require much longer than allowed times, the ignition



Fig. 10.

Calculated maximum pressure as a function of time. The pairs of numbers labeling the curves are ignition velocity (cm/µs) and initial surface-to-volume ratio (1/cm).

front may be considered better as the onset of rapid burning following ignition.

The required surface-to-volume ratios are small compared to the surface-to-volume ratios of the PETN powder used experimentally to make the explosive pressings. The powders used have (S/V)'s of 6200 and 15 000/cm. The value of (S/V) = 600/cm corresponds to that of 0.01-cm cubes and (S/V) = 60/cm corresponds to 0.1-cm cubes. Such particles are 10 to 20 times larger than the pressing powder particles. Pressing the powder to higher densities used in the donor charge eliminates much of the available surface of the original powder.

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## REFERENCES

- Robert H. Dinegar and Daniel T. Varley III, "All-Secondary Explosive Hot-Wire Devices," Los Alamos Scientific Laboratory report LA-7897-MS (October 1979).
- 2. Charles L. Mader, Numerical Modeling of Detonations (University of California Press, 1979).
- 3. Charles A. Forest, "Burning and Detonation," Los Alamos Scientific Laboratory report LA-7245 (July 1978).
- B. N. Kondrikov, V. M. Raikova, and B. S. Samsonov, "Kinetics of the Combustion of Nitro Compounds at High Pressures," Combustion, Explosion, and Shock Waves 9, 68-73 (1974) (a Plenum Pub. Corp. translation of Fizika Goreniya I Vzryva).

- Charles L. Mader and Charles A. Forest, "Two-Dimensional Homogeneous and Heterogeneous Detonation Wave Propagation," Los Alamos Scientific Laboratory report LA-6259 (June 1976).
- 6. J. B. Ramsay and A. Popolato, "Analysis of Shock Wave and Initiation Data for Solid Explosives," in *Fourth Symposium (International) on Detonation*, White Oak, Maryland, Office of Naval Research Symposium report ACR-126 (1966).
- Charles L. Mader and Milton Samuel Shaw, "User's Manual for SIN," Los Alamos Scientific Laboratory report LA-7264-M (September 1978).
- John M. Walsh and Russell H. Christian, "Equation of State of Metals from Shock Wave Measurements," Phys. Rev. 97, 1544 (1955).
- Charles L. Mader, "FORTRAN BKW: A Code for Computing the Detonation Properties of Explosives," Los Alamos Scientific Laboratory report LA-3704 (July 1967).

- Charles L. Mader, "Detonation Properties of Condensed Explosives Computed Using the Becker-Kistiakowsky-Wilson Equation of State," Los Alamos Scientific Laboratory report LA-2900 (July 1963).
- 11. Dante Stirpe, James O. Johnson, and Jerry Wackerle, "Shock Initiation of XTX-8003 and Pressed PETN," J. Appl. Phys. 41, 3884 (1970).
- G. E. Seay and L. B. Seely, Jr., "Initiation of a Low-Density PETN Pressing by a Plane Shock Wave," J. Appl. Phys. 32, 1092 (1961).
- 13. Terry R. Gibbs and Alphonse Popolato, Eds., LASL Explosive Property Data, (University of California Press, 1980).
- Bart Olinger and Howard Cady, "The Hydrostatic Compression of Explosives and Detonation Products to 10 GPa (100 kbars) and Their Calculated Shock Compression: Results for PETN, TATB, CO<sub>2</sub>, and H<sub>2</sub>O," in Sixth Symposium (International) on Detonation, Office of Naval Research Symposium report ACR-221 (1976), pp. 700-709.

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